ASTE 580 (Orbital Mechanics I) - Summer 2022

Course Syllabus

Instructor: Dr. Ryan Park
Class Location: OHE136
Time: Monday and Wednesday, 6:40-9:20 PM
E-Mail: Ryan.S.Park@jpl.nasa.gov
Please include “ASTE580” in the subject.
Course Website: https://courses.uscden.net
Teaching Assistant: TBD
Office Hours/Location/Phone Number: TBD
Tuesday 4:30 - 6:00 PM PST
Wednesday 4:30 - 6:00 PM PST
Thursday 4:30 - 6:00 PM PST
Remote Access: https://usc.zoom.us/j/99301316653?pwd=S2JIWDRUdS9FSU15eHILdkR5N3FJUT09

Required Text


Prerequisites

Graduate standing in engineering or science
Proficient in programming (e.g., MATLAB, PYTHON)
ASTE 480 (Spacecraft Dynamics) recommended

Course Objective

Orbital mechanics is the basis for spacecraft mission design and navigation and is a key component of spacecraft engineering and operations. The basic principles of orbits and astrodynamics inform the designer with options for selecting orbits, maneuvers, and mission profiles that impact the eventual spacecraft design. Understanding orbit perturbations, trajectories, and maneuver needs guides the mission planner with selecting optimal orbit maintenance, rendezvous, and transfers to accomplish the ultimate goals of the mission. This information
will then be used by the spacecraft and subsystem engineers to ensure that the spacecraft design can satisfy those requirements and achieve mission success. Once a spacecraft is on orbit, orbital mechanics is the foundation for tracking, orbit determination, and computing orbit corrections.

The goal of this course is to provide the student with an understanding of the basic theory, practices, and applications of orbit mechanics. It combines the foundational principles of algebra, geometry, and physics to describe the motion of objects to, in, and from space.

At the end of the course, the student should have mastered the basic principles of an object in orbit around a central body, effects of other forces on objects in orbit, and the means for changing orbits. These fundamentals will enable the student to learn and master almost any orbit analysis related problem that they will encounter during their professional lives.

**Course Topics**

This course covers standard concepts and methods applicable to practical and realistic astrodynamics problems. Topics include: the two-body problem, Keplerian orbits, the \(N\)-body problem, transfer orbits, planetary equations of motion, and numerical integration. Other topics as time permits.

**Grading**

- Homework (total of 6): 35%
- Midterm: 30%
- Final: 35%

**Weekly schedule**

**Note:** All homework assignments must be submitted by 6:00 pm on the due date (for both DEN and on-campus students). All homework assignments must be submitted via the course website as a single file (PDF preferred). **Late submission will not be accepted.**

- 05/18 (W): Lecture 01
- 05/23 (M): Lecture 02
- 05/25 (W): Lecture 03, Homework 1 due
• 05/30 (M): No class (Memorial Day)

• 06/01 (W): Lecture 04, Homework 2 due

• 06/06 (M): Lecture 05, Homework 3 due (likely no class, in which case, a make-up class will be given on 6/2)

• 06/08 (W): Midterm exam, 7-9 pm (open book; extra 30 minutes will be given for downloading and uploading, i.e., due time is 9:30 pm)

• 06/13 (M): Lecture 06, Homework 4 due

• 06/15 (W): Lecture 07, Homework 5 due

• 06/20 (M): Lecture 08, Homework 6 due

• 06/22 (W): Lecture 09, Homework 7 due

• 06/27 (M): Lecture 10, Homework 8 due

• 06/28 (T): Final exam, 7-9 pm (open book; extra 30 minutes will be given for downloading and uploading, i.e., submission due time is 9:30 pm)

References


